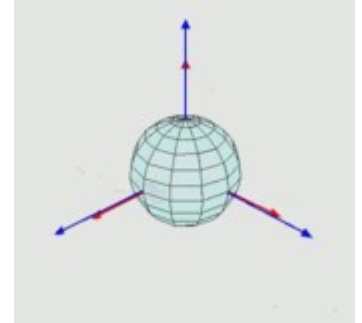


ROTATIONAL KINEMATICS

Angular Velocity Vector

- Consider a rotation which evolves over time:

$$\hat{x}'(t), \hat{y}'(t), \hat{z}'(t) \quad \text{or} \quad R(t) = \begin{pmatrix} R_{xx}(t) & R_{xy}(t) & R_{xz}(t) \\ R_{yx}(t) & R_{yy}(t) & R_{yz}(t) \\ R_{zx}(t) & R_{zy}(t) & R_{zz}(t) \end{pmatrix}$$



- Both axis and rate of rotation may change over time

- Over infinitesimal time interval between t and $t+dt$:

- Rotation about an “instantaneous axis” by **small** angle $d\theta$:
- $R_{\vec{A}}(d\theta) = R_x(d\theta_x) R_y(d\theta_y) R_z(d\theta_z)$ (in any order)
- Can write **rate** of rotation as a vector – xyz treated equally

- Angular velocity** vector:

$$\vec{\Omega} \equiv \frac{d\theta_x}{dt} \hat{i} + \frac{d\theta_y}{dt} \hat{j} + \frac{d\theta_z}{dt} \hat{k}$$

(Points
along axis
of rotation)

Rotational Kinematics

- Taylor Series for angular velocity:

$$\vec{\Omega} = \vec{\Omega}_0 + \left. \frac{d \vec{\Omega}}{dt} \right|_{t_0} (t - t_0) + \frac{1}{2} \left. \frac{d^2 \vec{\Omega}}{dt^2} \right|_{t_0} (t - t_0)^2 + \dots$$

“angular acceleration” ($\vec{\alpha}$) – may or may not point along $\vec{\Omega}$

- Similar to translational kinematics, with no “position vector”

- For rotations about a **constant** axis:

- Rotations do commute → can assign an “angular position” θ
- Taylor Series for rotation angle (**about a constant axis only**):

$$\theta = \theta_0 + \left. \frac{d \theta}{dt} \right|_{t_0} (t - t_0) + \frac{1}{2} \left. \frac{d^2 \theta}{dt^2} \right|_{t_0} (t - t_0)^2 + \dots$$

1-D kinematics equations for x, v_x, a_x can now be applied to θ, Ω, α

Time Derivative in Rotating Frames

- R matrix transforms components of a vector: $\vec{u}' = R \vec{u}$
 - Time derivatives in rotating frames must take into account:
 - 1) time dependence of the **actual** vector
 - 2) changing direction of **coordinate axes** → time dependence of R

- Example: Frame S' rotating with angular velocity $\vec{\Omega}$

- Using the **chain rule**: $\frac{d \vec{u}'}{dt} = (R_{\vec{\Omega}}) \frac{d \vec{u}}{dt} + \left(\frac{d R_{\vec{\Omega}}}{dt} \right) \vec{u}$

- By definition: $\frac{dR_{\vec{\Omega}}}{dt} = \frac{R_{\vec{\Omega}}(t + dt) - R_{\vec{\Omega}}(t)}{dt} = \left(\frac{R_{\vec{\Omega}}(\Omega dt) - 1}{dt} \right) R_{\vec{\Omega}}(t)$

Plug in $R_{\vec{\Omega}}(\Omega dt) = \begin{pmatrix} 1 & \Omega_z dt & -\Omega_y dt \\ -\Omega_z dt & 1 & \Omega_x dt \\ \Omega_y dt & -\Omega_x dt & 1 \end{pmatrix} \longrightarrow \frac{dR_{\vec{\Omega}}}{dt} = \begin{pmatrix} 0 & \Omega_z & -\Omega_y \\ -\Omega_z & 0 & \Omega_x \\ \Omega_y & -\Omega_x & 0 \end{pmatrix} R_{\vec{\Omega}}$

Time Derivative in Rotating Frames

- Examine the effect of matrix $\begin{pmatrix} 0 & \Omega_z & -\Omega_y \\ -\Omega_z & 0 & \Omega_x \\ \Omega_y & -\Omega_x & 0 \end{pmatrix}$ on a vector

$$\begin{pmatrix} 0 & \Omega_z & -\Omega_y \\ -\Omega_z & 0 & \Omega_x \\ \Omega_y & -\Omega_x & 0 \end{pmatrix} \begin{pmatrix} b_x \\ b_y \\ b_z \end{pmatrix} = \begin{pmatrix} \Omega_z b_y - \Omega_y b_z \\ \Omega_x b_z - \Omega_z b_x \\ \Omega_y b_x - \Omega_x b_y \end{pmatrix} = \boxed{-\vec{\Omega} \times \vec{b}}$$

The cross product can be written as an operator in matrix form!

- Time derivative of a vector in a rotating frame:

$$\frac{d\vec{u}'}{dt} = R \frac{d\vec{u}}{dt} - [\vec{\Omega} \times (R\vec{u})] \longrightarrow \text{Vectors which are defined by a time derivative (e.g. velocity) pick up an extra term when they are transformed}$$

- Examples: Velocity $\rightarrow \boxed{\vec{v}' = R\vec{v} - [\vec{\Omega} \times (R\vec{r})]}$

- Acceleration $\left\{ \begin{aligned} \frac{d\vec{v}'}{dt} &= R \frac{d\vec{v}}{dt} + \frac{dR}{dt} \vec{v} - \left[\vec{\Omega} \times \left(R \frac{d\vec{r}}{dt} + \frac{dR}{dt} \vec{r} \right) \right] \\ \vec{a}' &= R\vec{a} - 2[\vec{\Omega} \times R\vec{v}] + [\vec{\Omega} \times (\vec{\Omega} \times R\vec{r})] \end{aligned} \right.$

Coriolis and Centrifugal Acceleration

$$\vec{a}' = R \vec{a} - 2 [\vec{\Omega} \times R \vec{v}] + [\vec{\Omega} \times (\vec{\Omega} \times R \vec{r})]$$

- More useful to relate **a'** to **v'** and **r'** (instead of **v** and **r**)
 - This way, all measurements can be made in rotating frame

Plug in $R \vec{r} = \vec{r}'$

and $R \vec{v} = \vec{v}' + [\vec{\Omega} \times \vec{r}']$

$$\vec{a}' = R \vec{a} - 2 [\vec{\Omega} \times (\vec{v}' + (\vec{\Omega} \times \vec{r}'))] + [\vec{\Omega} \times (\vec{\Omega} \times \vec{r}')]$$

$$\vec{a}' = R \vec{a} - 2 [\vec{\Omega} \times \vec{v}'] - [\vec{\Omega} \times (\vec{\Omega} \times \vec{r}')]$$

Inertial acceleration

Coriolis acceleration

Centrifugal acceleration

$\mathbf{a}_{\text{centrifugal}}$ (Index Notation)

$$\vec{a}_{\text{centrifugal}} = -[\vec{\Omega} \times (\vec{\Omega} \times \vec{r})] = -\Omega_i \hat{e}_i \times (\Omega_j \hat{e}_j \times r_k \hat{e}_k)$$

$$\vec{a}_{\text{centrifugal}} = -\Omega_i \hat{e}_i \times (\Omega_j r_k \epsilon_{jkl} \hat{e}_l) = -\Omega_i (\Omega_j r_k \epsilon_{jkl}) (\epsilon_{ilm} \hat{e}_m)$$

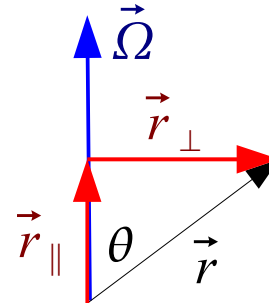
$$\vec{a}_{\text{centrifugal}} = (-\epsilon_{jkl} \epsilon_{mil}) \Omega_i \Omega_j r_k \hat{e}_m = (\delta_{ji} \delta_{km} - \delta_{jm} \delta_{ki}) \Omega_i \Omega_j r_k \hat{e}_m$$

$$\vec{a}_{\text{centrifugal}} = \Omega_i \Omega_i r_k \hat{e}_k - \Omega_i \Omega_j r_i \hat{e}_j$$

$$\vec{a}_{\text{centrifugal}} = \Omega^2 \vec{r} - (\vec{\Omega} \cdot \vec{r}) \vec{\Omega}$$

$$\vec{a}_{\text{centrifugal}} = \Omega^2 \vec{r} - (\Omega r \cos \theta) (\Omega \hat{\Omega})$$

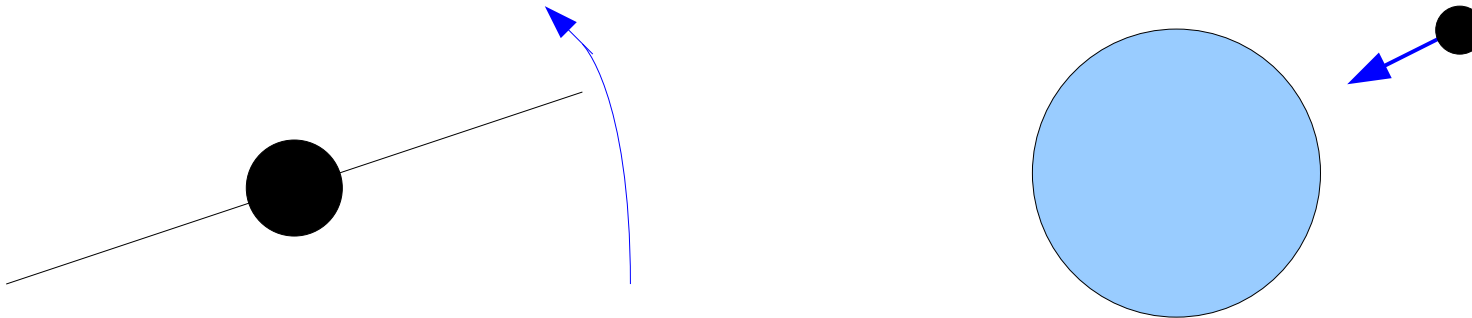
$$\vec{a}_{\text{centrifugal}} = \Omega^2 (\vec{r} - \vec{r}_{\parallel}) = \boxed{\Omega^2 \vec{r}_{\perp}}$$



- $\mathbf{a}_{\text{centrifugal}}$ always points directly **away** from rotation axis

Coriolis / Centrifugal Examples

- Bead on spoke of bicycle wheel rotating at Ω
 - Slides outward such that $r(\theta) = r_0 + k(\theta - \theta_0)$ (ignore gravity)
 - Calculate $r'(t)$, $v'(t)$, Coriolis and centrifugal accelerations



- Spacecraft approaches S.B. from above (34° latitude)
 - At some instant:
 - Earth rotates at angular velocity Ω , spacecraft is at $r = 2R_E$
 - Falling straight down at speed v (in inertial space)
 - Calculate (North/South, East/West, Up/Down) components of v'
 - Calculate components of Coriolis and centrifugal accelerations

Fictitious vs. Real Forces

- Normal, Friction, Tension, Compression
 - Exerted by **particles** on other particles nearby
 - Newtonian Physics does not attempt to describe further
- Does a particle exert Coriolis and centrifugal forces?
 - **No!** They are a property of the rotating **space** itself
- Which type of force is gravity? Has aspects of both...
 - Gravitational force is directly exerted by space
 - But has particles as its source
 - Einstein's work shows gravity acts more like a **fictitious** force
 - Examples: 1) Does gravity affect light? 2) orbit of Mercury